Multidisciplinary and Multilevel Aircraft Design Methodology using Enhanced Collaborative Optimization

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Since aircraft design is an inherently multidisciplinary undertaking, the quest for increasingly optimized solutions can only be comprehensively successful by implementing multilevel or system design optimization architectures to step up disciplinary optimizations. In this study, a methodology which uses the Enhanced Collaborative Optimization (ECO) multilevel architecture with the purpose of developing a multidisciplinary design optimization methodology within the context of the preliminary design stage of unmanned aerial vehicles is presented. The concepts of weighting coefficient and dynamic compatibility parameter are presented and assessed for the ECO architecture. A routine that calculates the aircraft performance for the mission profile and vehicle's performance metrics under consideration has been implemented using low fidelity models for the aerodynamics, stability, propulsion, weight, balance and flight performance. A benchmarking case study of two different mission profiles for evaluating the advantage of using a variable span wing within the optimization methodology developed is also featured.

Nomenclature

= wingspan, m = win mean chord, m C_l = airfoil lift coefficient = propeller diameter, inch d_{prop} = Oswald efficiency number E_{to} = take-off energy, J E_{cb} = climb energy, J E_{cz} = cruise energy, J E_{dt} = descent energy, J ECO = enhanced collaborative optimization FW= fixed wing = minimum altitude for each mission stage, m h_{min} = maximum altitude for each mission stage, m h_{max} h_{to} = take-off altitude for each mission stage, m MDO = multidisciplinary design optimization Opt. = optimization = propeller pitch, inch p_{prop} = range, m RoC= rate-of-climb, m/s **VSW** = variable span wing V= velocity, m/s = wind velocity, m/s V_{wind}

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